

P.O. BOX 3090 ONE AYRES WAY ALBANY, GEORGIA 31706-3090 PHONE 229/883-1440 FAX 229/439-9790

Service Bulletin No. SB-AG-19 January 20-1987 Page 1 of 3

FUSELAGE UPPER LONGERON INSPECTION AND REPAIR

MODELS AFFECTED: Model S2R, S/N 1416R thru 2574R

> Model S2R-R1340, S/N 1340-001 thru R1340-010 Model S2R-R3S, S/N R3S-001 thru R3S-010

REASON FOR

PUBLICATION: Preclude possible fatigue failure of the upper fuselage longeron.

COMPLIANCE: Part I: Within next 25 hours in service.

Part II: Prior to next flight if cracks are found.

BY WHOM WORK WILL

BE ACCOMPLISHED: A & P Mechanic

APPROVAL: **FAA Approved**

ESTIMATED

MAN HOURS: Part I: 15 Minutes

Part II: 50 Man Hours

SPECIAL TOOLS: NONE

ACCOMPLISHMENT INSTRUCTIONS:

Part I:

- 1. Remove the fuselage side skin located between the wing upper surface and the hopper. This will provide access to the upper longeron from the firewall to the cockpit.
- 2. Visually inspect the longeron paying particular attention to: (See Figure 1)
 - 1. Area around the upper engine mount attach points.
 - 2. Each vertical diagional tube intersection at the upper longeron.

- 3. Each hopper attachment bracket.
- 3. If no cracks are found, fill out and mail Compliance Card noting that no cracks were found. Proceed to Record Compliance and return aircraft to service.
- 4. If cracks are found, fill out and mail Compliance Card noting that cracks were found, then proceed to Part II

PART II.

- 1. If any cracks are found, remove hopper and attaching parts. Weld cracks closed in longerons. Obtain Custom Kit No. CK-AG-19 from Ayres Corporation.
- 2. Grind off the four 10866-1 Attach Brackets, See Figure 2.
- 3. Grind or nibble the ends of the sleeve tubes P/N CK-AG-19-1 and –2 to fit snuggly within the original weld beads at the forward tube clusters/gussets at the firewall. See Figure 3.
- 4. Clamp the split sleeves in position and weld as shown in Figure 2.
- 5. Grind the fiberglass in the hopper saddles as necessary to allow the hopper to be lowered onto the longerons in the original hopper location.
- 6. Locate and weld the forward attach brackets 2 inches aft of its original location. Locate and weld the aft attach brackets at its original location (plus or minus ½ inch). Use a straight edge resting on the remaining (aft) brackets to obtain lateral alignment of the replacement brackets.
- 7. Paint the repaired area.
- 8. Reinstall hopper and attaching parts.

ELECTRICAL LOAD: No Charge

WEIGHT AND BALANCE: No Change

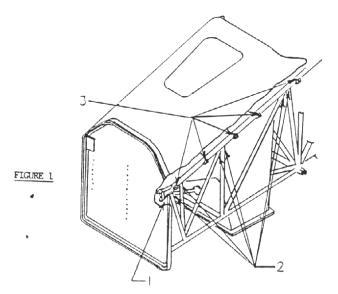
RECORD COMPLIANCE: Make appropriate entry in aircraft records as follows: Service

Bulletin No. SB-AG-19 dated January 20, 1987, entitled "Fuselage Upper Longeron Repair," accomplished Part I (date), Part II (date) (if

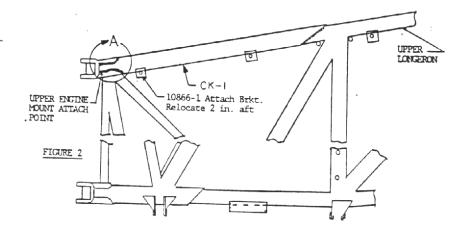
applicable).

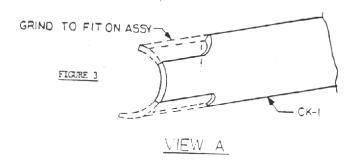
PARTS LIST:

PART NUMBER	DESCRIPTION
CK-AG-19-1	Split Sleeve
CK-AG-19-2	Split Sleeve
10866-1	Attach Brackets
	CK-AG-19-1 CK-AG-19-2



- 1. UPPER ENGINE MOUNT ATTACH POINT
- 2. VERTICAL AND DIADONAL TUBE INTERSECTION
- 3. HOPPER ATTACHMENT BRACKETS





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